

Space Legos: A Concept for In-Space Assembly of Large Structures with a Stationary Robot

Thesis by
Sahangi Dassanayake

In Partial Fulfillment of the Requirements for the
Degree of
Doctor of Philosophy in Space Engineering

The logo for the California Institute of Technology (Caltech), featuring the word "Caltech" in a bold, orange, sans-serif font.

CALIFORNIA INSTITUTE OF TECHNOLOGY
Pasadena, California

2025
Defended September 30, 2024

© 2025

Sahangi Dassanayake
ORCID: 0000-0002-1363-5764

All rights reserved

ACKNOWLEDGEMENTS

This thesis would not have been possible without the support, encouragement, and guidance of many individuals who have shaped my journey, both academically and personally.

First and foremost, I would like to express my deepest gratitude to my advisor, Professor Sergio Pellegrino. Your insightful guidance, passion for the field, and unwavering support have been pivotal throughout my Ph.D. journey. Thank you for encouraging me to explore my ideas independently, for keeping me focused when it was most needed, and for the many exciting discussions and opportunities you have generously provided.

I would like to express my heartfelt thanks to my collaborators in the DARPA NOM4D project at the Space Structures Lab. Jaden, Alan, and Charles—your mentorship, wisdom, and shared enthusiasm for innovation have enriched my research experience. You encouraged my ideas, fostered collaboration, and without your dedication, this thesis would not have taken the same form. I also wish to extend my sincere thanks to Mark Thomson, as well as Professor Igor Bargatin and his team at the University of Pennsylvania, for their invaluable guidance and contributions throughout this work. A special thank you to Dr. Andrew Detor for his leadership of the DARPA NOM4D program and for supporting this research through grant HR001122C0054.

I am sincerely thankful to my candidacy and thesis committee members, Professors Domniki Asimaki, Guruswami Ravichandran, Ares Rosakis, and especially, Daniel Meiron. Your feedback, encouragement, and support have been vital in helping me navigate this academic journey.

To my dear labmates, thank you for being there to exchange ideas and share the highs and lows of graduate school (and endure the many practice talks!). I'm incredibly fortunate to have had such a supportive and fun group by my side. Every conference, lab party, and group meeting was brighter and filled with unforgettable memories, thanks to you.

To my GALCIT/MCE G1 cohort—Sorina, Niyati, Tracy, Stephanie, John, Brayden, Paul, Tanner, Miles, Peter, Ying, Nikhil—our shared experiences have been a cornerstone of my time here. You made the tough times (G1 and Quals during

COVID!!) easier and the good times even better, and I'm grateful for every moment of fun, laughter, and friendship we have had along the way.

A special thanks to the entire GALCIT community and the dedicated administrative staff for your tireless efforts in keeping everything running so smoothly behind the scenes. I deeply appreciate all that you do.

To my wonderful friends in the performing arts—especially Nancy, your creativity and passion for your craft have always been a source of inspiration. You welcomed me with open arms and reminded me that art and science are intertwined in their pursuit of truth and beauty, and I am lucky to have shared this journey, and the stage, with you.

To my Caltech friends—Shruti, André, Scott, Yanky, Hirsh, Taylan, Utku, and Mert, since the very beginning of grad school, you have been there for me, celebrating my successes and offering encouragement through the setbacks. Your warmth, kindness, and unfaltering support have brought me so much comfort and joy. I moved to LA knowing no one, and I am incredibly grateful to you for making this city feel like home. I'm so fortunate to have found such an incredible community here.

To my friends who have supported me from across continents—Roanga and Nipun, your constant presence, despite the distance and time zone differences, has meant more to me than I can express. You listened to my endless rants, offered thoughtful advice, and shared in both my frustrations and triumphs. Your unwavering encouragement and friendship have been a grounding force throughout this journey, and I'm incredibly grateful to have had you by my side, even from afar.

Finally, to my family—Amma, Appachchi, Nangi—I don't have enough words to express my gratitude. Amma and Appachchi, you instilled in me curiosity, confidence, and an eagerness to learn that has guided me throughout my life. You gave me the freedom to make mistakes, while always assuring me of your pride in my aspirations and your unconditional love and support—even going so far as to arrange for home-cooked meals to be sent when I was too busy to eat. Nangi, thank you for being there to listen when I needed someone to talk to during my moments of anxiety. And to Bubbles, my dog, thank you for being a sweet and constant source of comfort. I love you all, and this thesis is as much yours as it is mine.

To everyone who has been a part of this journey, whether mentioned here or not, thank you from the bottom of my heart. Your presence in my life has made all the difference.

ABSTRACT

Human nature is inherently driven by the desire to build; advancing from primitive shelters to skyscrapers, and extending this relentless pursuit of progress to space through technological innovations. As space missions require larger and more complex structures, traditional deployable systems face challenges due to constraints on launch mass, volume, and complex deployment mechanisms. In-space assembly (ISA) offers a promising solution for constructing large structures, such as telescopes and satellites, directly in space.

This thesis introduces a novel ISA concept with a centralized ‘truss builder’ for autonomous assembly of polygonal-ring structures, using simple, repetitive operations and focusing on scalable mesh reflectors for communication and imaging. Utilizing the standard AstroMesh architecture, a rapid generalized design method is developed. Through the analysis of reflector geometry, optimized cable prestress, structural design, and a high-fidelity finite element model, analytical scaling laws are derived for mass, stowed envelope, and natural frequency based on aperture diameter. A semi-analytical homogenization model is introduced to efficiently predict fundamental natural frequencies. Stowed volume is a key limitation for large deployable reflectors, approaching current and future launch capacity limits, while the proposed ISA reflectors face no such constraints for apertures up to 200 meters.

A two-dimensional finite element model simulates the assembly kinematics of large ring-like structures with the proposed ISA concept, enhancing understanding of the process and evaluating key design aspects of a stationary robot assembling scalable ring-like trusses. The model provides insights for optimizing autonomous assembly systems and underscores the need for advanced numerical simulations to ensure smooth assembly and stability during ISA, especially as structures scale.

Lab-scale prototype testing validates the ISA concept, with results aligning qualitatively with simulations. Both experiments and simulations reveal a range of viable solutions, demonstrating flexibility for future mission designs. This research offers crucial insights into the design and scaling of mesh reflectors, setting the stage for comparing ISA with traditional deployable systems. The proposed ISA concept presents a practical solution for building high-precision, large-scale structures in space, advancing the field of space construction and supporting future extended space missions.

PUBLISHED CONTENT AND CONTRIBUTIONS

- [1] J. Suh, S. P. Dassanayake, M. Thomson, and S. Pellegrino, “Scaling Laws for Deployable Mesh Reflector Antennas,” *AIAA Journal*, pp. 1–12, 2024. DOI: 10.2514/1.J063940.
S.D. participated in conceptualizing the project, developed the generalized design method, performed the scaling analysis for mass and stowed volume and contributed to the preparation of the manuscript.
- [2] J. Suh, S. P. Dassanayake, M. Thomson, and S. Pellegrino, “In-Space Assembly of Large Mesh Reflector Antennas,” in *Aerospace Structures, Structural Dynamics, and Materials Conference, SSDM 2024 [Technical Presentation]*, 2024, p. 137 740.
S.D. participated in conceptualizing the project, developed the generalized design method and simulation techniques, and performed the scaling analysis for mass and stowed volume and the numerical analyses on assembly.
- [3] S. P. Dassanayake, J. Suh, M. Thomson, and S. Pellegrino, “Mass, Volume and Natural Frequency Scaling of Deployable Mesh Reflectors,” in *AIAA SCITECH 2024 Forum*, 2024, p. 2041. DOI: 10.2514/6.2024-2041.
S.D. participated in conceptualizing the project, developed the generalized design method, performed the scaling analysis for mass and stowed volume and prepared the manuscript.
- [4] J. Suh, S. P. Dassanayake, M. Thomson, and S. Pellegrino, “Scalable Concept for Reflector Antenna Assembled in Space,” in *AIAA SCITECH 2024 Forum*, 2024, p. 0823. DOI: 10.2514/6.2024-0823.
S.D. participated in conceptualizing the project and developed the generalized design method.
- [5] J. Suh, S. Dassanayake, M. Thomson, and S. Pellegrino, “Concept for In-Space Assembly of Large Reflector Antennas,” in *41st ESA Antenna Workshop ESTEC*, 2023.
S.D. participated in conceptualizing the project, developed the generalized design method, performed the scaling analysis and contributed to the preparation of the manuscript.
- [6] J. Suh, S. P. Dassanayake, and S. Pellegrino, “In-Space Assembly of Large Mesh Reflectors,” in *AIAA SCITECH 2025 Forum [Accepted]*, 2025.
S.D. developed the simulation techniques, performed the numerical analyses, and contributed to the preparation of the manuscript.
- [7] S. P. Dassanayake and S. Pellegrino, “In-Space Assembly of Large Structures with a Stationary Robot,” *[In Preparation]*, 2025.

S.D. developed the simulation techniques, performed the detailed numerical analyses and prepared the manuscript.

TABLE OF CONTENTS

Acknowledgements	iii
Abstract	v
Published Content and Contributions	vi
Table of Contents	vii
List of Illustrations	x
List of Tables	xv
Nomenclature	xvi
Chapter I: Large Structures in Space	1
1.1 In-Space Assembly: A Strategy for Launching Large Space Structures?	1
1.2 Robotic Autonomous Assembly	3
1.3 Reflector Antennas	6
1.4 Feasibility of In-Space Assembly Concepts	8
1.5 Thesis Outline	8
Chapter II: Establishing the Limits of Deployable Reflectors	11
<i>Article: Scaling Laws for Deployable Mesh Reflector Antennas</i>	
2.1 Motivation	12
2.2 Geometry, Connectivity, and Prestress	14
2.2.1 Geometry and Design for Kinematic and Static Determinacy	14
2.2.2 Prestress Optimization of Cable Nets	17
2.2.3 Potential Modifications to the Minimum Tension Requirement	20
2.3 Structural Design	22
2.3.1 Metallic Mesh and Cable Net Design	22
2.3.2 Perimeter Truss Design	22
2.3.3 Joint Design	23
2.4 Scaling of Mass and Volume	25
2.4.1 Estimating Mass and Stowed Volume	25
2.4.2 Mass and Stowed Dimensions Results	26
2.4.3 Analytical Scaling Laws	27
2.4.4 Scaling Studies for Deeper Reflectors	29
2.5 Scaling of Natural Frequencies of Vibration	31
2.5.1 Finite Element Model and Boundary Conditions	31
2.5.2 Natural Frequencies and Mode Shapes	32
2.5.3 Semi-Analytical Models for Fundamental Frequencies	34
2.6 Chapter Conclusions	39
Chapter III: ISA Concept for Ring-Like Structures	41
3.1 Motivation	42
3.2 Scaling of Mass and Stowed Volume for ISA Reflectors	43
3.3 In-Space Assembly Concept	45

3.3.1	General Overview	45
3.3.2	Operations of the Truss Builder	47
3.4	Chapter Conclusions	49
Chapter IV:	Simulation of ISA Concept	50
4.1	Motivation	51
4.2	Numerical Simulation Setup	52
4.2.1	Model Definition	53
4.2.2	Modeling the Cable Net	54
4.2.3	Simulation Steps	56
4.3	Tuning the Simulation for a Six-Sided Structure	59
4.3.1	Angle Stops	59
4.3.2	Assembly Plate Orientation	60
4.3.3	Damping Coefficients	61
4.4	Simulation Results for the Six-Sided Structure	62
4.5	Chapter Conclusions	65
Chapter V:	Simulation of Twelve-Sided Reflector	67
5.1	Motivation	67
5.2	Effect of Assembly Plate Orientation	68
5.3	Effect of Prestressing Method	70
5.4	Effect of Cable Net Orientation	72
5.5	Further Considerations for Assembly Simulation	75
5.5.1	‘Jamming’ Condition	75
5.5.2	Cable Activation and Bay Release Sequence	75
5.5.3	Prestressing at the Final Assembly Stage	76
5.6	Chapter Conclusions	77
Chapter VI:	Experimental Demonstration of ISA Concept	79
6.1	Motivation	80
6.2	Design of In-Space Assembly Facility	80
6.2.1	Lab-scale Reflector Prototype	80
6.2.2	Truss Builder Prototype	82
6.3	Design Considerations	84
6.3.1	Angle Stops and Assembly Plate Orientation	84
6.3.2	Prestressing Method	86
6.4	Chapter Conclusions	88
Chapter VII:	Summary and Perspectives	90
7.1	Summary	90
7.2	Future work	92
7.3	Perspectives	95
	An Illustration of ISA Reflectors: Can E.T. “phone home” after all?	96
	Bibliography	97
	Appendix A: Homogenized stiffness matrix of parallel tessellation	104
	Appendix B: Modification of cable stiffness in ABAQUS/CAE	105

LIST OF ILLUSTRATIONS

<i>Number</i>	<i>Page</i>
1.1 State-of-the-art deployable structures: (a) JWST (NASA, 2016) and (b) AstroMesh deployable mesh reflector (Northrop Grumman, 1999)	2
1.2 Lightweight trusses as support structures: (a) erectable 4-meter tetrahedral truss [17] and (b) JWST support structure (NASA, 2015)	3
1.3 Advancements in robotic assembly: (a) SpiderFab robot constructing a support structure for functional elements [9], Robot-Assembled Modular Space Telescope (RAMST) architecture [29]: dexterous hexbot (b) deploying and assembling truss and (c) traversing and assembling mirror module and BILL-E: robotic platform [34]: identical bipedal robots (d) using the truss as support for traversing, (e) placing and (f) moving identical truss modules.	5
1.4 Faceted reflective surface: (a) tension truss concept and (b) structural architecture of faceted deployable mesh reflector, based on [40].	7
2.1 Generalized design terminology: structural architecture of faceted deployable mesh reflector.	12
2.2 Geometry of paraboloidal reflector.	14
2.3 Faceting of paraboloidal surface: a) variation of facet size with diameter and b) schematic layout of net.	15
2.4 Three different net configurations with a) non-triangular facets, b) cross wires, and c) complete triangular tessellation.	16
2.5 Pin-jointed bar model of front net for $n = 3, n_t = 2, n_c = 2$.	16
2.6 Comparison of cable prestress distributions for $D = 200$ m, $F/D = 1.0$ corresponding to (a) 2 distinct tension tie forces, and (b) $(n + 1)$ rings of tension tie forces.	18
2.7 Definition of <i>rings</i> in inner net.	19
2.8 Cable net configuration for $D = 100$ m, $n_t = 13, n_c = 3$: a) original configuration, b) mapped configuration 1, and c) mapped configuration 2.	21
2.9 Perimeter truss: a) fully deployed and stowed configurations, and b) relative position of members in stowed configuration.	24

2.10	CAD images of a) Type-1 joint with 5 struts, b) Type-1 joint strut sleeves, c) Type-2 joint with 3 struts, and d) Type-2 joint strut sleeves.	24
2.11	Areal density of reflector components ($F/D = 1.0$): a) non-optimal prestress design, and b) optimal prestress design.	26
2.12	Scaling of deployable reflectors for $F/D = 1.0$: a) total mass, b) stowed diameter, and c) stowed height.	27
2.13	Process of establishing a scaling law for the total mass.	28
2.14	Scaling of deployable reflectors for $F/D = 0.5$ and 0.7 : a) total mass, b) stowed diameter, and c) stowed height.	29
2.15	Batten-supported boundary condition: a) prime batten, and b) boundary conditions in finite element model.	32
2.16	Natural frequencies and mode shapes for $F/D = 1.0$: a) frequency trends, b) first two mode shapes for prime-batten support condition, and c) first mode shape for free-free condition.	33
2.17	Natural frequency trends: a) $F/D = 0.5$, and b) $F/D = 0.7$	34
2.18	Semi-analytical modeling scheme.	35
2.19	Semi-analytical model to estimate lateral mode: a) 1-DoF model, and b) homogenization of the net.	36
2.20	Natural frequency corresponding to lateral mode.	37
2.21	Semi-analytical model for free-free saddle mode: homogenization of a) net, and b) perimeter truss.	38
2.22	Natural frequency of free-free saddle mode: a) frequency comparison, and b) mode shape comparison.	39
3.1	Coiled strut volume estimation: a) core mandrel, b) cross section of the strut, c) flattened strut, and d) strut spool.	44
3.2	Scaling of ISA vs. deployable reflectors for $F/D = 1.0$: a) total mass, b) stowed diameter, and c) stowed height.	44
3.3	ISA concepts: (a) linear ISA trusses [9], and (b) proposed two-dimensional ISA for ring-like structures with a cable interior.	45
3.4	Schematic drawing of in-space polygonal ring assembly with cable interior: (a) truss assembly, and (b) cable net deployment.	46
3.5	Two-dimensional view of truss assembly process: first, second, third, and final bay assembly.	48
4.1	Simulation approach: perimeter ring and a single cable net.	53

4.2	Finite element model definition for a) an example polygonal ring geometry, and b) corresponding initial configuration where rods and cables are modeled with their final connectivity.	54
4.3	Simplification of cable net: (a) six-sided structure, and twelve-sided structure: (b) simplification-1, and (c) simplification-2.	55
4.4	Specified nonlinear penalty function for cable stiffness in the finite element model.	56
4.5	Framework of simulation steps.	57
4.6	Boundary conditions when a) pushing-out bays B_1 and B_2 , b) activating cables connected to N_3 , c) pushing out bay B_3 , and d) activating cables connected to N_4	58
4.7	Geometry of six-sided structure.	59
4.8	Angle stops: a) geometry of the structure, b) cables transitioning from slack to taut during bay push out, and c) definition of angle stop.	60
4.9	Two assembly plate orientations: a) perpendicular to the truss builder face: $\theta = 90^\circ$, and b) tilted at angle $\theta < 90^\circ$	61
4.10	Cable extension results for the six-sided structure; $\theta = 90^\circ$	63
4.11	Cable extension results for the six-sided structure; $\theta = 60^\circ$	65
4.12	Intermediate polygonal shapes: (a) $\theta = 90^\circ$, (b) $\theta = 80^\circ$, and (c) $\theta = 60^\circ$	66
5.1	Geometry of twelve-sided reflector.	67
5.2	Relationship between critical assembly plate orientation, θ_{cr} and interior angle of perimeter truss.	68
5.3	Effect of assembly plate orientation: (a) $\theta = 90^\circ$, (b) $\theta = 60^\circ$, and (c) $\theta = 30^\circ$	69
5.4	Two-dimensional view of the modified prestressing method: moving truss support, i^{th} bay assembly and prestressing the structure.	71
5.5	Effect of the prestressing method: $\theta = 60^\circ$, (a) moving last bay: S_1 is fixed (b) moving truss support: S_1 is moved by 0.1 m at the start and reset at the end of assembly.	72
5.6	Two configurations of the cable net: (a) orientation 1 (b) orientation 2.	73
5.7	Effect of cable net orientation relative to the truss support: prestressing by moving last bay and at $\theta = 60^\circ$, (a) orientation 1 (b) orientation 2.	73

5.8	Effect of cable net orientation relative to the truss support: prestressing by moving truss support and at $\theta = 60^\circ$, (a) orientation 1 (b) orientation 2.	74
5.9	'Jamming' condition for $\theta = 60^\circ$, orientation 2 and prestressing by moving last bay: (a) 'jammed' configuration, and (b) resolved configuration.	75
5.10	Cable extension results for the twelve-sided reflector: $\theta = 60^\circ$, orientation 1 and prestressing by moving last bay, (a) original cable stiffness definition, and (b) modified cable stiffness definition to reflect prestress.	76
6.1	Lab-scale demonstration: a) target reflector, and b) dimensions of struts.	81
6.2	Reflector prototype, $D = 1.4$ m: a) CAD drawing of the joint and joint-strut attachment, b) prototypes of the joint, strut, and a single bay, and prototype of cable net with c) a push-latch device installed at the outer node of cable net and attached to the truss, and d) a tension tie.	82
6.3	Truss builder prototype: a) schematic drawing of truss builder and components, and prototypes of: b) truss builder and c) manipulator.	83
6.4	Reflector assembly demonstration for $\theta = 90^\circ$: a) initial state, b) $i = 3$, c) $i = 5$, d) $i = 10$ with kink formation, and e) $i = 12$: completed reflector.	85
6.5	Comparison of experimental and simulation results for $\theta = 60^\circ$: a) $i = 9$, b) $i = 11$, and c) $i = 12$: completed reflector.	86
6.6	Relationship between θ and working space for the robotic manipulator.	86
6.7	Reflector prototype assembled at $\theta = 72^\circ$, prestressed by moving the last bay: a) collinear truss nodes, and b) undesirable stretching of cables.	87
6.8	Assembly demonstration for $\theta = 72^\circ$, prestressed by moving the truss support: a) $i = 3$, b) truss support S_1 moved, c) $i = 7$, d) $i = 10$, e) $i = 12$: prior to repositioning S_1 , and f) $i = 12$: completed reflector, overlaid with the corresponding simulation result.	88

- 7.1 Perforated Kapton films, 25.4 μm thick, with a 100 nm aluminized coating on one side for reflectivity: a) a single facet of cable net, with a magnified view of the perforation, patterned using a CO_2 laser, and b) achievable reduction of mass with lowered biaxial prestress, σ for $F/D = 1.0$ 93
- 7.2 Potential enhancements to improve packaging efficiency: (a) deployable coilable Omega boom [73], and (b) additive manufacturing of components in space [74]. 94

LIST OF TABLES

<i>Number</i>	<i>Page</i>
2.1 Comparison of Objective Functions for $D = 200$ m, $F/D = 1.0$. . .	20
2.2 Comparison of cable net configurations for $D = 100, 200$ m, $F/D = 1.0$	21
2.3 Prestress range (Nets: tension; Truss: compression), for $F/D = 1.0$.	22
2.4 Size of perimeter truss members, for $F/D = 1.0$	23
2.5 Joint masses for $F/D = 1.0$	25
2.6 Joint masses for $F/D = 0.5, 0.7$	29
2.7 Natural frequency of lateral mode (high-fidelity model vs. semi-analytical model)	37
2.8 Natural frequency of saddle mode (high-fidelity model vs. semi-analytical model)	39
4.1 Sensitivity to assembly plate orientation and damping coefficient . . .	62
B.1 Material definition for a single cable	105
B.2 Material definition for cable i in a multi-cable system	105

NOMENCLATURE

Space Legos. Modular components engineered for assembling structures in space.

Stationary Robot. A robot that remains fixed in position relative to the spacecraft and does not traverse across or along the structure being assembled.